#### Broadband Reflective Coating Process for Large FUVOIR Mirrors

NASA SBIR Phase II contract No. NNX14CG39P Technical Monitor: Dr. Manuel Quijada (JPL)



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#### FUVOIR Coating Performance Objectives

**Table 1- Performance Objectives** 

Metric	SBIR Goal			
Reflectance (90-nm to 110-nm)	>60%			
Reflectance (110-nm to 200-nm)	> 85%			
Reflectance (200-nm – 2500-nm)	> 90%			
Surface Roughness	<5 A RMS			
Coating Stress	< 85 MPa			
Humidity	95% RH, 50 C, 24 hour			
Moderate Abrasion	20 rub, 5 psi, cheese cloth			
Thermal Cycling	-80 + 50 C (ten cycles)			
Adhesion	ASTM Tape Test			

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### "There is nothing new except what has been forgotten"

-Marie Antoinette

#### Oxidation of aluminum film in 5x10^-7 torr vacuum



R.P. Madden, L.R. Canfield, and G. Hass, "On the Vacuum-Ultraviolet Reflectance of Evaporated Aluminum before and during Oxidation", *Journal of the Optical Society of America* **Vol. 53 No. 5** (1963)

# FUSE: Lessons learned for future FUV missions

- FUSE: SiC coatings 905 A 1,105 A, Al:LiF 987 A 1,187 A.
- LiF-coated mirrors were exposed to 30%- 50% RH for less than 5-days during ground storage yet the mirrors degraded prior to launch from 70% reflectance to 55% reflectance
- Time required to degrade un-oxidized bare aluminum in the vacuum of space
  - A 3 months to 2 years in low earth orbit (to drop to 35%, which is approximately the reflectance of SiC)
  - 20-years or more in higher orbit (L2) (provided outgassing from spacecraft doesn't kill the reflectivity)
- Al/MgF2 reflectance ~15% 90-nm to 100-nm

### Summary: Types of Contamination

- Oxygen
  - Contaminates bare aluminum
    - Ground storage
    - In the vacuum of space, spacecraft outgassing, (LEO worse)
- Water
  - Humidity during ground storage
  - Outgassing from spacecraft
- Organic
  - Ground storage
  - Outgassing from spacecraft (UV exposure from sun in space makes this worse)

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What things will we need to coat large mirrors with Manuel's 3-step process?

- A technique to heat a surface to ~250 C and apply a coating, without heating the entire mirror assembly (may not need this step if Sn over-coat idea works)
- A method to apply very thin layers (2-nm) uniformly over large areas
- A way to protect the sensitive fluoride layers from humidity during ground storage (apply Sn and remove in space by heating; 230 C melt temperature)

#### Use ZeCoat's moving source technology to apply a very thin layer, quickly over a large mirror



Radiant heating during metal-fluoride deposition heats aluminum coating to 250 C before entire mirror assembly gets hot



#### Tin (Sn) as a protective cover for LiF or Al?

#### 449.5°F (231.9°C)

Tin, Melting point



For bare aluminum in space facing sun  $\alpha/\epsilon \sim 0.07/0.03 = 2.33$ Temperature = 217 C

#### Tin

Chemical Element

Tin is a chemical element with symbol Sn and atomic number 50. It is a main group metal in group 14 of the periodic table. Wikipedia

Symbol: Sn

Electron configuration: Kr 4d10 5s2 5p2

Melting point: 449.5°F (231.9°C)

Atomic number: 50

Electrons per shell: 2, 8, 18, 18, 4

Discovered: 3500 BC

Atomic mass: 118.71 u

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#### FUV Deposition and Monitoring Set-Up





#### **Deuterium lamp and PMT**

#### PHOTOMULTIPLIER TUBE R1081



Figure 2: Typical Gain Characteristics



SUPPLY VOLTAGE (V)

#### Upward view looking at sample holder and quartz crystal monitor



# FUV coating set-up with moving substrate holder



# Pyrometer (8-14 $\mu$ ) canister with zinc selenide window



1-2 micron pyrometer can look through glass window into vacuum chamber





#### Resistive (3) source heat-sink





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### Phase I Challenges

- Demonstrating uniform, 1-nm coating over large area
- Demonstrating complete removal of Sn from Surface
- Measuring temperature of coated aluminum surface
  - Pyrometer very sensitive to stray light in chamber

### Transmission (%) vs Wavelength (nm) measured at (4) radial positions

NiCr 1 pass



NiCr 3 passes



## Phase I results (coating uniformity over large area using motion control evap. system)

NiCr films with varying thickness



# Conversion of transmission data to layer thickness using optical constants n,k

#### Average thickness (nm)

	(nm)								
	Layers	1	2	3	5	6	8	8	
	3	1.51	3.01	4.25	6.90	8.18	10.38	10.67	
Radial	16	1.80	3.17	4.83	7.43	8.85	11.41	11.56	
Position (cm)	33	1.49	2.80	4.38	6.70	8.17	10.38	10.51	
	49	1.52	2.74	4.22	6.50	7.88	10.16	10.16	

#### Average thickness per layer (nm)

				(nm)					
	Layers	1	2	3	5	6	8	8	Avg
	3	1.51	1.51	1.42	1.38	1.36	1.30	1.33	1.40
Radial	16	1.80	1.59	1.61	1.49	1.48	1.43	1.45	1.55
Position (cm)	33	1.49	1.40	1.46	1.34	1.36	1.30	1.30	1.38
	49	1.52	1.37	1.41	1.30	1.31	1.27	1.27	1.35
	Avg	1.58	1.46	1.47	1.38	1.38	1.32	1.34	1.42

### To be discussed in final report....

- Sn removal experiments (reflectance before and after removal)
- Surface heating experiments (ability to produce crystalline fluorides by heating surface and producing large temperature gradient between the front and back of the mirror)

#### QUESTIONS?